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Effect of Fabrication Methods on Static Strength of Polymer Based Composites under the Low Temperature Range

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Key Words: Wind Turbine Blade (), Vacuum Bagging Process (),

RIM (Resin Injection Molding), Composites ()

Abstract

When the wind turbine is used in cold regions, the mechanical properties and dimension stability of the blade will be changed. The proposal of this paper is to test the durability of the blade for wind turbine. It is necessary to select the most comfortable materials and fabrication processes for more stable wind turbine blade in cold regions. To select the most comfortable materials and processes, the static strength has to know through the tensile static tests at the severe condition as cold regions. First, the tensile static specimens made by RIM (Resin injection molding) process & vacuum bagging process with reinforcement materials and resin. Tensile static tests were carried out on three laminate lay-ups (carbon prepreg, carbon fiber dry fabric and glass fiber dry fabric) at different test temperature(24 °C -30 °C determining properties such as the mechanical strength, stiffness and strain to failure. At different test temperature, in order to test the tensile strengths of these specimens used the low temperature chamber. Next, the results of this test were compared with each other. Finally, the most comfortable materials and fabrication processes can select based on these results. The results show the changes in the static behavior of three laminate lay-ups at different test temperatures. At low temperatures, the static strengths are higher than the ones at room temperature.

wet lay-up 가 가 가 wet lay-up (f) (e) prepreg vacuum infusion 가 (h) (g)가 가 , 가 (i)Fig. 1 RIM process 가 RIM(Resin injection molding) (peel ply) (resin flower) (sealant 2. 시편제작 tape) (vacuum bag film) 2.1 RIM process 가 가 (a) (P) RIM 가 2.2 Vacuum bagging process Vacuum bagging process (c)(d)

RIM

vacuum bagging process fabrication Fig. 3

2.3 시편

 $(183 \text{mm} \times 19 \text{mm})$

ASTM D638-9 (Fig. 4)
Tensil kut (dog bone)

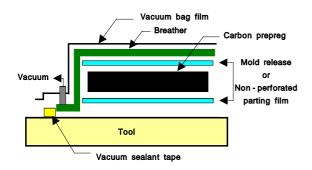


Fig. 2 Vacuum bagging process

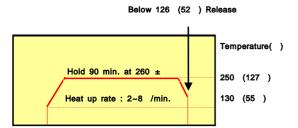


Fig. 3 Cure cycle for vacuum bagging system

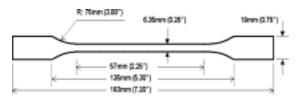


Fig. 4 The shape of specimen (ASTM D638-9)

Table 1 Overview of materials and fabrication method

Group	Reinforcement		Resin			Test	
	Type	Volume %	Type	Volume %	Process	Method	
A	Carbon	56.6%	F	43.4%	Vacuum	Tensile	
	prepreg	30.0%	Epoxy	43.4%	Bagging		
В	Carbon						
	fiber dry	45.7%	Epoxy	54.3%	RIM	Tensile	
	fabric						
С	Glass fiber	59.6%	Epoxy	40.40/	RIM	Tensile	
	dry fabric	39.6%		40.4%	KIN	Lensiie	

Table 1 vacuum bagging process RIM process

volume %
. () TB 社 C3327 dry
fabric prepreg, 社 G635
dry fabric , dry fabric
prepreg 社
YD-115, RIM

G-A0432, D-230 vacuum bagging process prepreg

G-A0432 D-230 50:50 , YD-115 100:35

3. 실 험

3.1 Test set up

MTS

, controller (Fig. 5). -30

3.2 Test

tensile strength Young's modulus ultimate strain



Fig. 5 Test set up()





(a) Before the test (b) After the test Fig. 6 Test set up()

glass dry fabric carbon dry fabric **RIM** vacuum bagging carbon prepreg 3가 process (-30)cross head speed 1.27mm/min.(0.05in/min.)

4. 결과 및 고찰

4.1 Static tensile test

Fig. 7 Fig. 8

carbon prepreg

carbon dry fabric 가

, glass 가 dry fabric

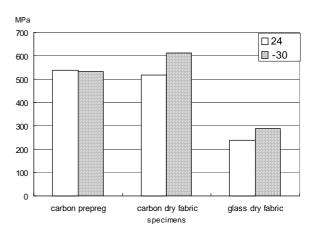


Fig. 7 Comparison of strength

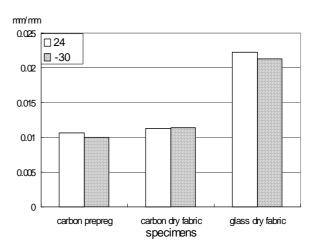


Fig. 8 Comparison of strain

carbon fiber fabrication method prepreg vacuum bagging system 가 . Table 1 vacuum bagging system RIM process . RIM process 가 가 breather

> carbon dry fabric RIM process

. RIM process vacuum bagging system	1 가 brittle 가	,				
4.2 SEM analysis Static tensile test			(a) 24 Fig. 9 Fracto	(tography of glast	o) -30 ss dry fabri	ic
SEM(Scanning electron n . static tensile test .	nicroscopy)	SEM				
(end)			() 21	-	(1) 20	
pull-out . ,	pull-out 가		(a) 24 Fig. 10 Fracto	ography of cart	(b) -30 oon dry fab	oric
Fig. 9 24	-30 matrix가		- , j.	- ¢	/ Ç *	
ductile n	matrix가 natrix	가	~~		de la	4
matrix	brittle	. ,		300	197	200
	matrix		(a) 24 Fig. 11 Frac	tography of ca	(b) -30	να
,	matrix		rig. 11 mac	lography of ca	roon prepre	g
가 pull-c						
	end)가			,	matrix	가
가			가	brittle		가
Fig. 10 Fig. 11 matrix	가 24					
, -30			(2)	carbon	prepreg	carbon
matrix			dry fabric		71	
, 가	matrix가 brittle		가		가	
matrix		glass dry fabric		가		
가 pull-out			glass fiber	/r brittle		
<i>5</i> -	경 로		•			
J. 1	결론		(3) vacuum baggii	ng system		carbon
(1)			RIM pro	ocess	ca	rbon
(1) static tensile test	carbon prepreg					vacuum
, carbon dry fabric	가 glass dry fabric		bagging			

vacuum bagging system

가 가

vacuum bagging system

.

(4) SEM

matrix가

matrix

pull-out

, glass dry fabric 가

가

(5) carbon fiber glass fiber

, carbon fiber glass fiber

glass

fiber carbon fiber

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