

# Simulation of Neutral Flow around Plasma Actuator

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## ABSTRACT

Numerical simulations were carried out of the effects of momentum and heat produced by a plasma actuator on neutral flow. Momentum and heat generated during plasma discharge were modeled as a body force and heat source using results of experiments and DSMC of particle. These force and heat model were inserted into a Navier-Stokes equation and the flow around the plasma actuator could be explored by solving fluid dynamics only. Fluid simulation showed that force produced in DSMC generated a jet flow in the vicinity of the plasma actuator and heat accounted for density change.

## INTRODUCTION

Plasma actuator have demonstrated the capability to apply to boundary layer control of separated flow of airfoil with high angle of attack<sup>1</sup>, low-pressure turbine blade<sup>2</sup>, and vortex control of body with blunted trailing edge<sup>3</sup>, etc. Experiments and simulations have shown that plasma actuator imparts momentum to the boundary layer flow and promote boundary layer attachment.<sup>4,5</sup> Figure 1 illustrates the configuration of plasma actuator, which is a rather simple and a device of the dielectric barrier discharge configuration with two electrodes, one buried and the other exposed to air. The

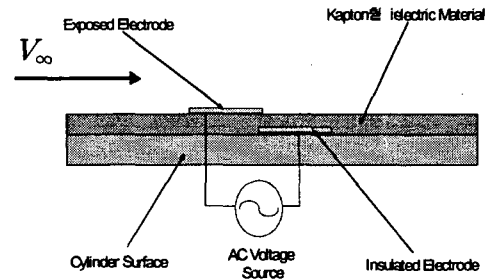


Figure 1. Configuration of plasma actuator

electrodes are subjected to an alternating current of several thousands volts at a frequency in up to tens of kilohertz.

## FORCE AND HEAT MODEL

Figure 2. shows a temporal distribution of force during one cycle of 5kHz alternating current applied to the plasma actuator. The initial discharge produces a negative force and the second discharge produces a positive force acting as thrust. The total integrated force was scaled to

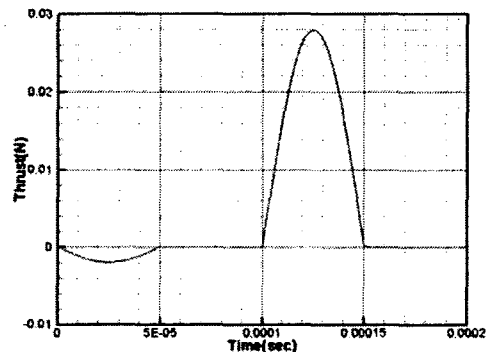


Figure 2. Temporal Distribution of Force produced by Plasma Actuator

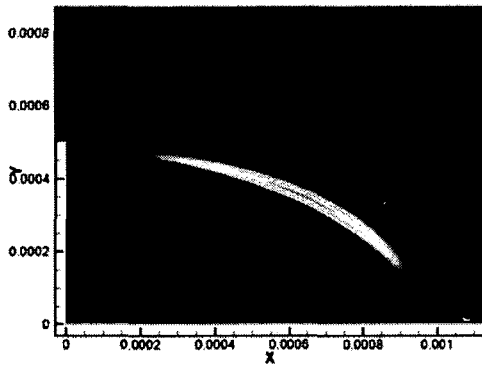


Figure 3. Spatial Distribution of Force produced by Plasma Actuator

produce an average force of about 1.5mN, which is representative of the experiments.

Spatial distribution of force was determined from the results of DSMC of air molecule during discharge<sup>6</sup>. Domain, where plasma exist and accelerate, is shown in figure 3 and force are concentrated in the center region.

## NUMERICAL ANALYSIS

In order to analyze the neutral flow around the plasma actuator, this modeling approach for force and heat produced by plasma actuator is implemented in the

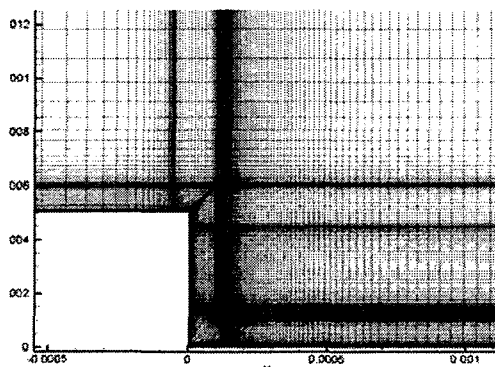


Figure 4. Grid near the edge of the exposed electrode

FLUENT. Grid for the computation domain, 40 cm long and 20 cm high, is shown in Figure 4 and flow was assumed to be laminar.

The force and heat was applied as a momentum and heat source to air which was not moving initially. A 2nd order time-accurate Newtonian sub-iteration scheme was applied to get unsteady feature of air flow. 200 time steps were put in one cycle of 5 kHz alternating current so the size of time step had to be 1 $\mu$ s. Even with 2D simulation, parallel computing were needed because millions of iteration should be carried out to reach the converged status of unsteady solution.

Figure 5. shows velocity magnitude of time-averaged solution during one cycle at 45ms after plasma actuator initiation. Temperature distribution near the edge of the exposed electrode is shown in figure. 6, which shows temperature rise of about 20K.

## REFERENCES

- 1) Martiqua L. Post and Thomas C. Corke, "Separation Control on High Angle of Attack Airfoil using Plasma Actuators," AIAA 2003-1024, 41st

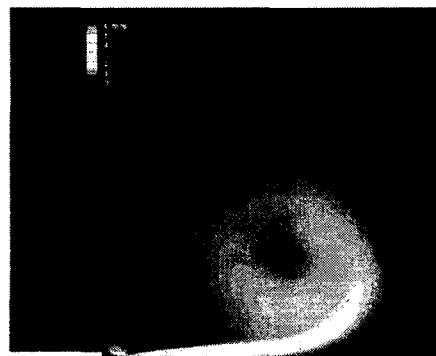


Figure 5. Velocity Magnitude Contours at 5kHz

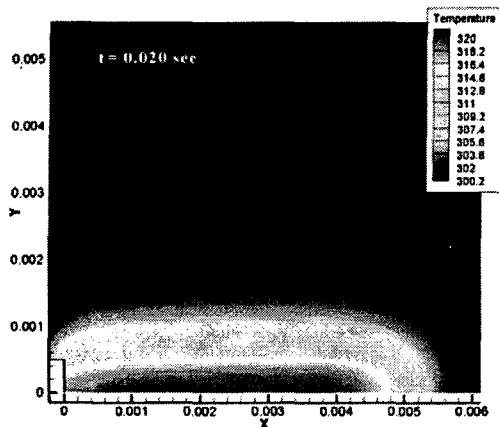


Figure 6. Temperature Distribution for  
10W at 5kHz

Aerospace Science Meeting and Exhibit,  
Jan. 2003

2) Junhui Huang, Thomas C. Corke, and  
Flint O. Thomas, "Plasma Actuators for  
Separation Control of Low Pressure  
Turbine Blades," 41st Aerospace Science  
Meeting and Exhibit, Jan. 2003

3) Thomas E. McLaughlin, Matthew D.  
Munsk, Joseph P. Vaeth, Travis E.  
Dauwalter, Jeffrey R. Goode, and Stefan  
G. Siegel, "Plasma-Based Actuator for  
Cylinder Wake Vortex Control," AIAA  
2004-2129, 2nd AIAA Flow Control  
Conference, Jun. 2004

4) C. L. Enloe, Thomas E. McLaughlin,  
Robert D. VanDyken, K. D. Kachner, Eric  
J. Jumper, and Thomas C. Corke,  
"Mechanisms and Responses of a Single  
Dielectric Barrier Plasma," AIAA  
2003-1021, 41st Aerospace Science  
Meeting and Exhibit, Jan. 2003

5) G. I. Font, S. Jung, C. L. Enloe, T. E.  
McLaughlin, W. L. Morgan, and J. W.  
Baughn, "Simulation of the Effects of  
Force and Heat Produced by a Plasma  
Actuator on Neutral Flow Evolution,"  
AIAA 2006-167, 44th AIAA Aerospace  
Science Meeting and Exhibit, Jan. 2006

6) Gabriel I. Font, "Boundary Layer  
Control with Atmospheric Plasma  
Discharge," AIAA 2004-3574, 40th  
AIAA/ASME/SAE/ASEE Joint Propulsion  
Conference and Exhibit, Jun. 2004