

(Electro-mechanical Battery)
(Flywheel Battery)

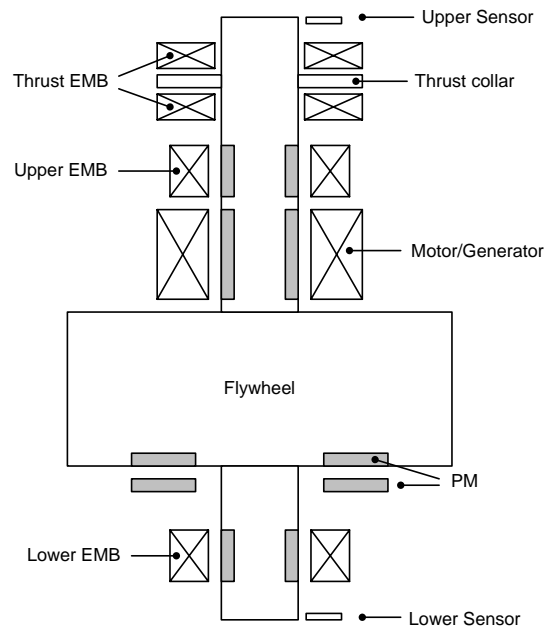


Fig. 1. Schematic diagram of a flywheel energy storage system showing its components.

[1].

2.2

2.2.1

8-pole

(x,y)

[2].

[3].

$$F_{\max} = \frac{A_g B_{\text{sat}}^2}{\mu_0} \quad (1)$$

Pole 8.8 cm²
1000N 가

, 1.2T 가

$$\left| \frac{dF}{dt} \right|_{\max} = F_{\text{sync}} \Omega_{\max} \quad (2)$$

20,000rpm 가
3 N/μsec

$$\left. \frac{dF}{dt} \right|_{\max} = \frac{2\alpha I_{\max} V_{\max}}{g_0} \quad (3)$$

Fig.2

2.2.2

2.3

가

가

가

가 1

[4].

가

$$F_z = \frac{\mu_0 A_g N^2}{4} \left[\frac{(I_b + i_p)^2}{(g_0 - z)^2} - \frac{(I_b - i_p)^2}{(g_0 + z)^2} \right] \quad (4)$$

$$K_z = - \left. \frac{\partial F_z}{\partial z} \right|_{z=0, i_p=0} = - \frac{\mu_0 A_g N^2 I_b^2}{g_0^3} \quad (5)$$

3G 가 (1)

250kg , pole 6421mm² ,

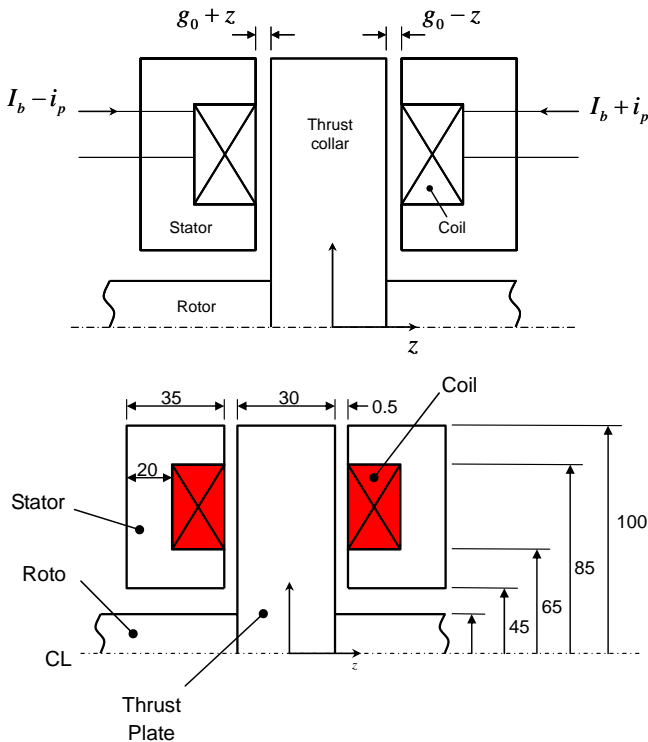


Fig. 2. Thrust bearing design (not scaled). The number of coil turns is 50, which satisfy the thermal and saturation requirements.

(6)-(10) Fig. 3

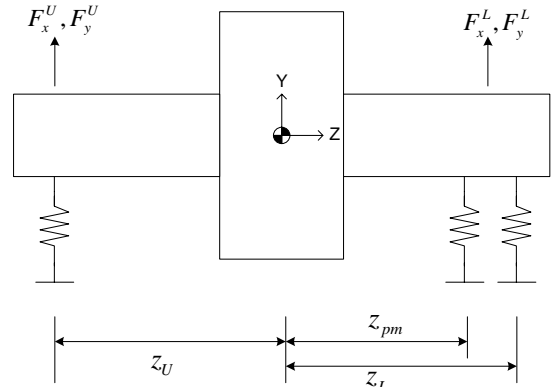


Fig. 3. Coordinate convention used in the dynamic modeling.

$$m\ddot{x} = -(2k_m + k_{pm})x - [k_m(z^U + z^L) - k_{pm}z^{PM}] \theta_y + k_i(i_{px}^U + i_{px}^L) \quad (11)$$

$$m\ddot{y} = -(2k_m + k_{pm})y + [k_m(z^U + z^L) - k_{pm}z^{PM}] \theta_x + k_i(i_{py}^U + i_{py}^L) \quad (12)$$

$$I_t \ddot{\theta}_x = k_m(z^U + z^L + z^{PM})y - k_m(z^{U^2} + z^{L^2} + z^{PM^2}) \theta_x - I_p \Omega \dot{\theta}_y + k_i(-z^U i_y^U + z^L i_y^L) \quad (13)$$

$$I_t \ddot{\theta}_y = -k_m(z^U + z^L + z^{PM})x - k_m(z^{U^2} + z^{L^2} + z^{PM^2}) \theta_y + I_p \Omega \dot{\theta}_x + k_i(z^U i_x^U - z^L i_x^L) \quad (14)$$

K_m

K_i

$$K_m = -\frac{\mu_0 A_g N^2 I_b^2}{g_0^3} \quad (15)$$

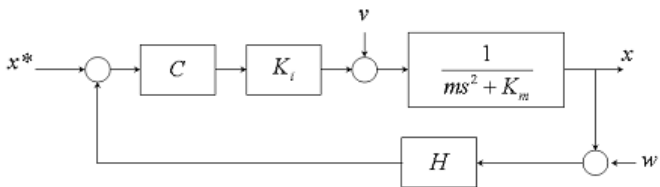
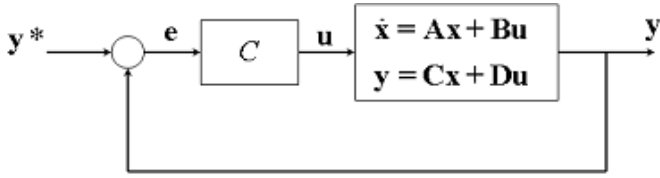


Fig. 4. Block diagram of the dynamic model and the reduced-order model.

$$K_i = \frac{\mu_0 A_g N^2 I_b^2}{g_0^2} \quad (16)$$

(11)-(14)

$$\begin{aligned} \dot{x} &= \mathbf{A}(\Omega)\mathbf{x} + \mathbf{B}\mathbf{u} \\ \mathbf{y} &= \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u} \end{aligned}$$

3.

(open-loop)
가

PID

. PID

gyroscopic

(translational motion)

(yaw and tilt motion)

$$m\ddot{x} = F \quad (17)$$

$$m\ddot{x} = -K_m x + K_i i_p + F_e \quad (18)$$

block diagram

Fig.

4

. Fig. 4

C

H

PD(Proportional-Derivative)

$$C(s) = K_p + K_D s \quad (19)$$

가

$$C(s) = K_p + \frac{K_D s}{\tau_1 s + 1} \quad (20)$$

$$C(s) = K_p \frac{\tau_2 s + 1}{\tau_1 s + 1} \quad (21)$$

τ_2

$$\tau_2 = \tau_1 + \frac{K_D}{K_p} \quad (22)$$

1

$$H(s) = \frac{K_s \omega_c}{s + \omega_c} \quad (23)$$

가

가

4.

Fig. 5

20,000 rpm

가

가

pole

. Fig. 5

pole

가

가 가

Gyroscopic

가

가

가

, pole

damping

. Fig. 6

1

(translation

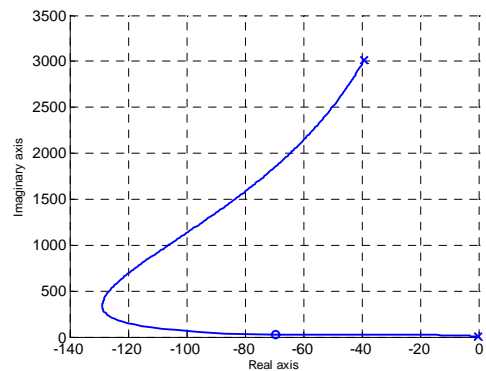


Fig. 5. Pole location of the system from zero to maximum speed mode)

(conical mode) 2 가 가

(forward whirl)

(backward whirl)

. Fig. 6

. Fig. 7

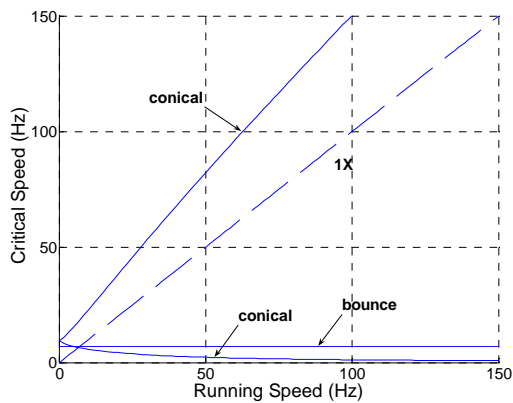


Fig. 6. Rigid body modes with respect to the running speed (simulation).

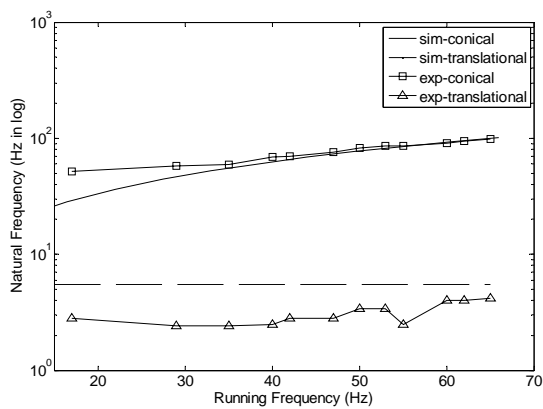


Fig. 7. Experimentally observed rigid body modes.

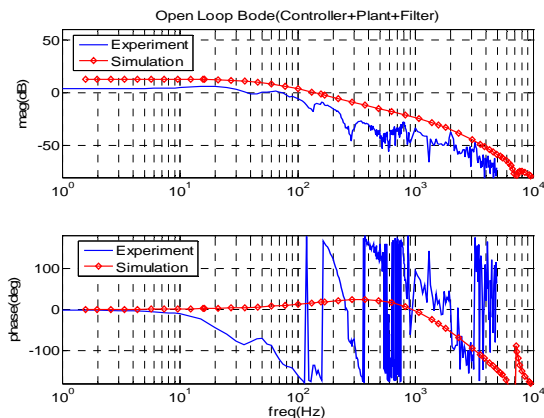


Fig. 8. Transfer function Bode plot obtained from simulation and experiment.

Fig. 8

Bode
Bode

가

Amplifier

5.

가

가

Gyroscopic
[5],

PID

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- (2) E. H. Maslen and D. C. Meeker, "Fault tolerance of magnetic bearings by generalized bias current linearization," *IEEE Trans. Magnetics*, vol. 31, pp. 2304-2314, May 1995.
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